

Abstract

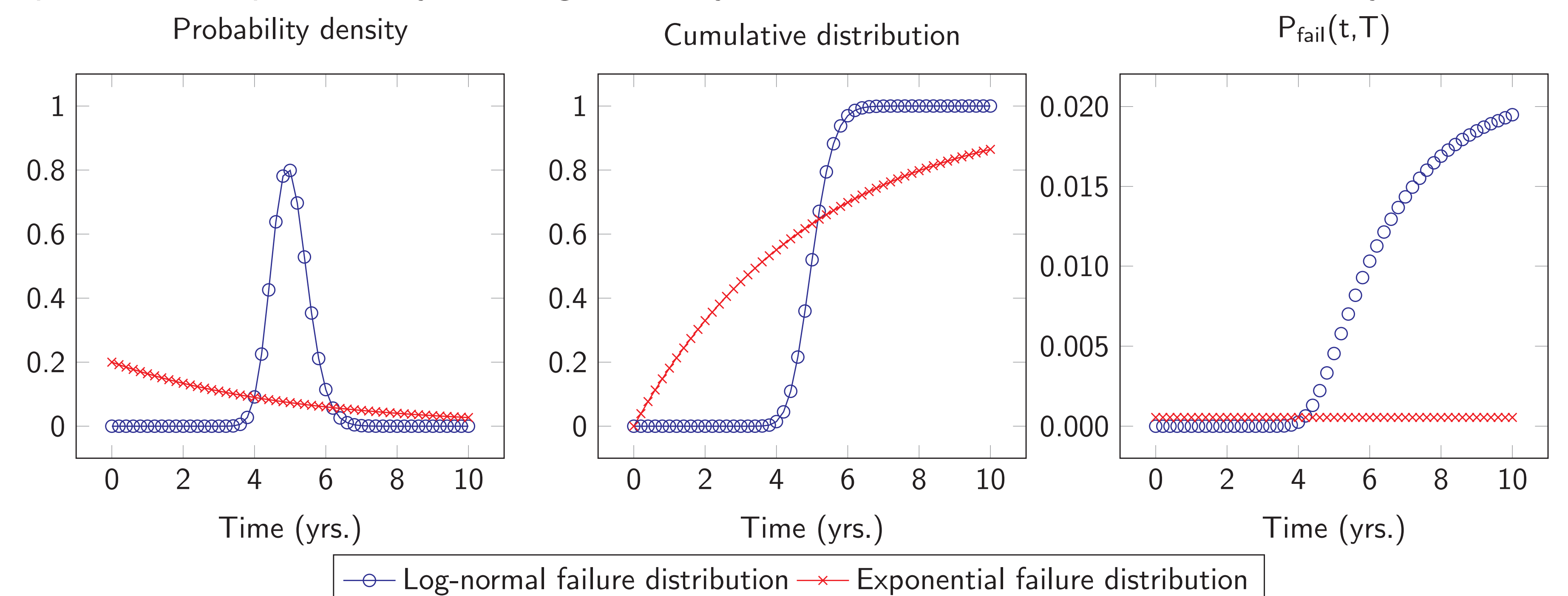
- Electronic components in space applications are subject to high levels of ionizing and particle radiation, causing transient errors and permanent failure.
- Transient errors can be detected and corrected using memory scrubbing. However, this causes an overhead that reduces both the availability and the lifetime of the system.
- We present a mechanism based on dynamic hidden Markov models that balances availability and lifetime by estimating and predicting the probability of the occurrence of permanent and transient faults.
- Results show that our model leads systems to their desired lifetime, and it outperforms rule-based and traditional hidden Markov models approaches.

3. Permanent Faults

The probability that a permanent fault will occur by time t , given that the system was still functional at the end of the previous scrubbing period of length T , is $P_{fail}(t, T)$.

- The memoryless exponential probability distribution is frequently used in the literature.
- Because memorylessness neglects wear, we adopt a log-normal failure distribution.

Comparison of exponentially and log-normally distributed failures with MTTF = 5 yrs.



1. Motivation

In space, integrated circuits are exposed to:

- High level of radiation.
- Heavy ion impacts.

These factors can disrupt circuits' behavior. Methods for the detection/recovery of transient errors exist [1], but there are no clear guidelines on how to deal also with permanent faults. Typical approaches consist of:

- Repeated computation.
- A threshold on consecutive errors [2].

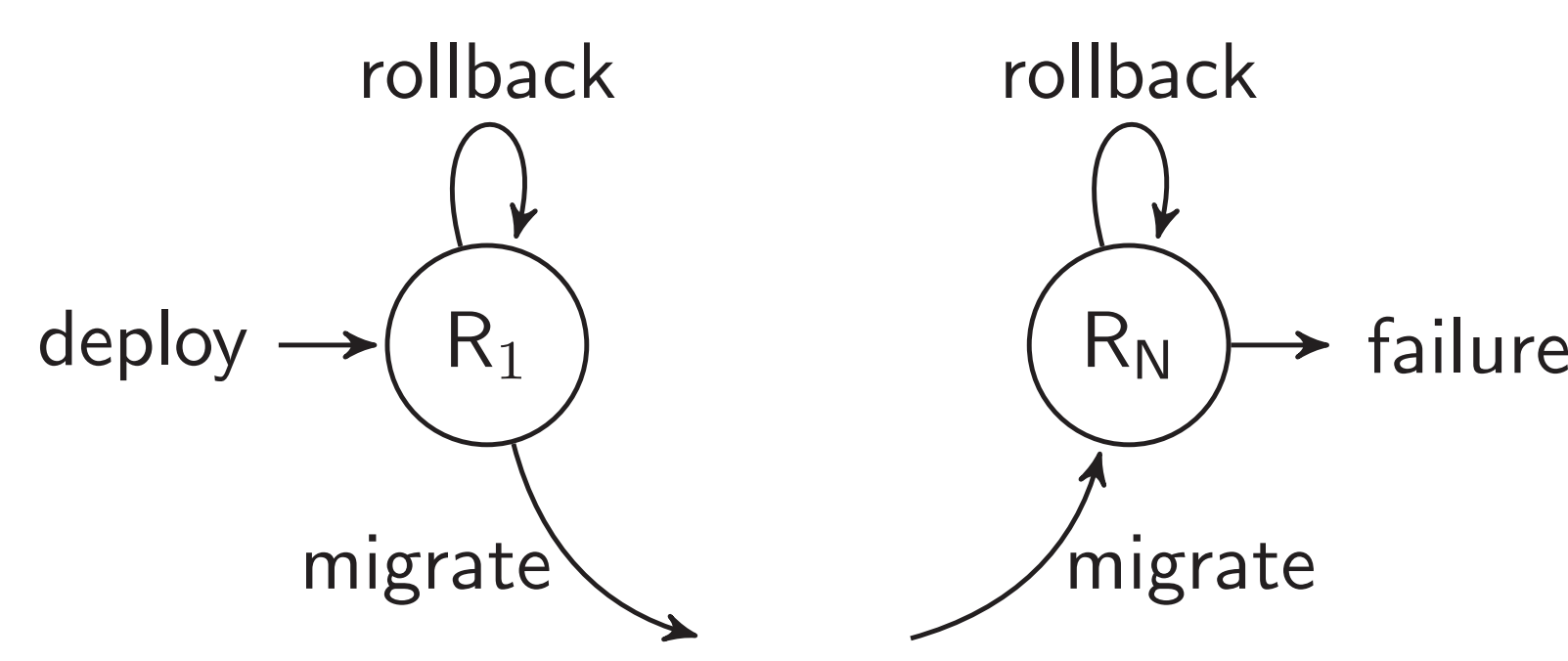
However, this methodology requires additional resources, reduces system availability, and accelerates the wear of the electronics.

4. Computing Model

The computing model abstracts:

- Homogeneous multi-core systems.
- Any N-times modular redundant system.

Hastily declaring failures reduces *lifetime*.
Lingering for too long reduces *availability*.

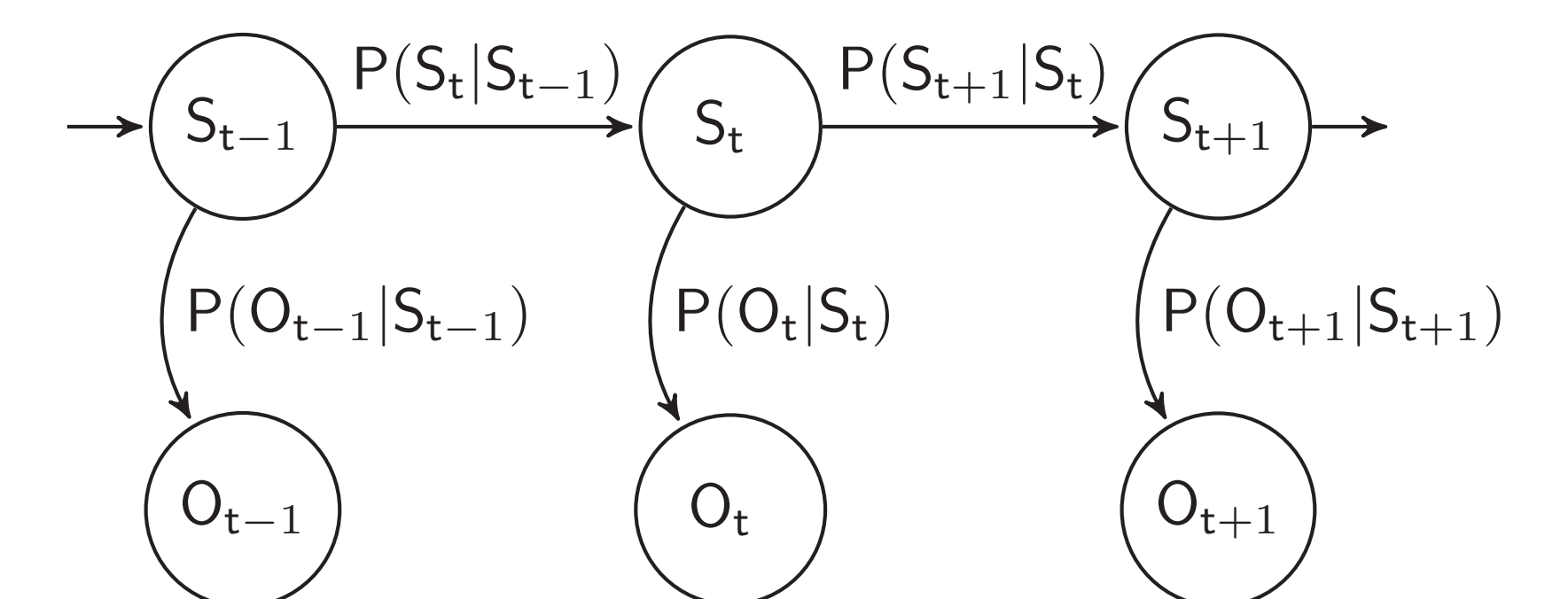


5. Hidden Markov Models

A mathematical framework to model:

- Discrete stochastic systems.
- Evolving stochastic systems.
- Partial-observability/imperfect sensors.

HMMs can predict the next system status.



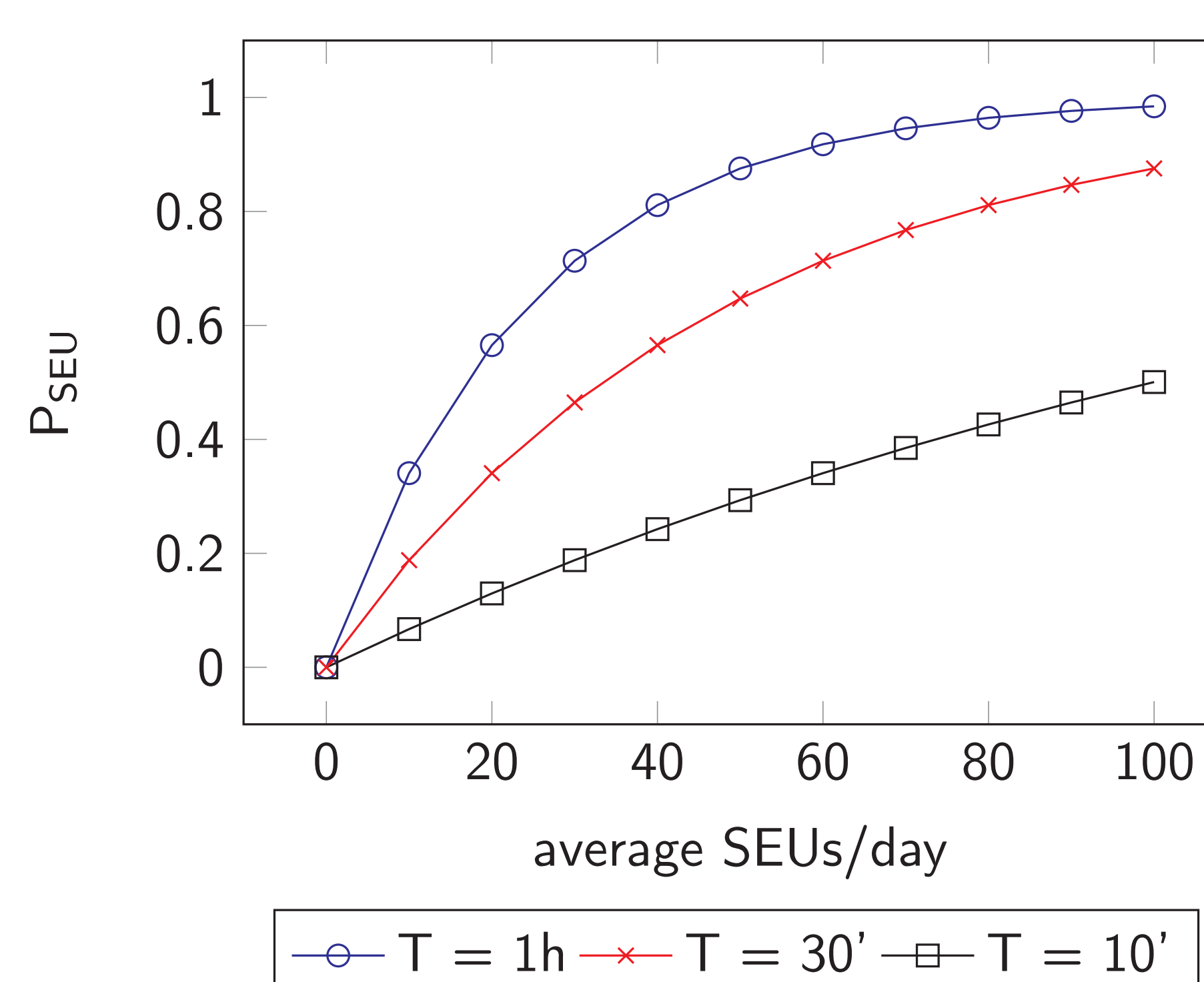
2. Single Event Upsets

We use probability theory to model the occurrence of faults.

SEUs are caused by high-energy particles:

- Whose impacts are independent.
- Which happen at a constant average rate.
- The rate is mission phase-dependent.

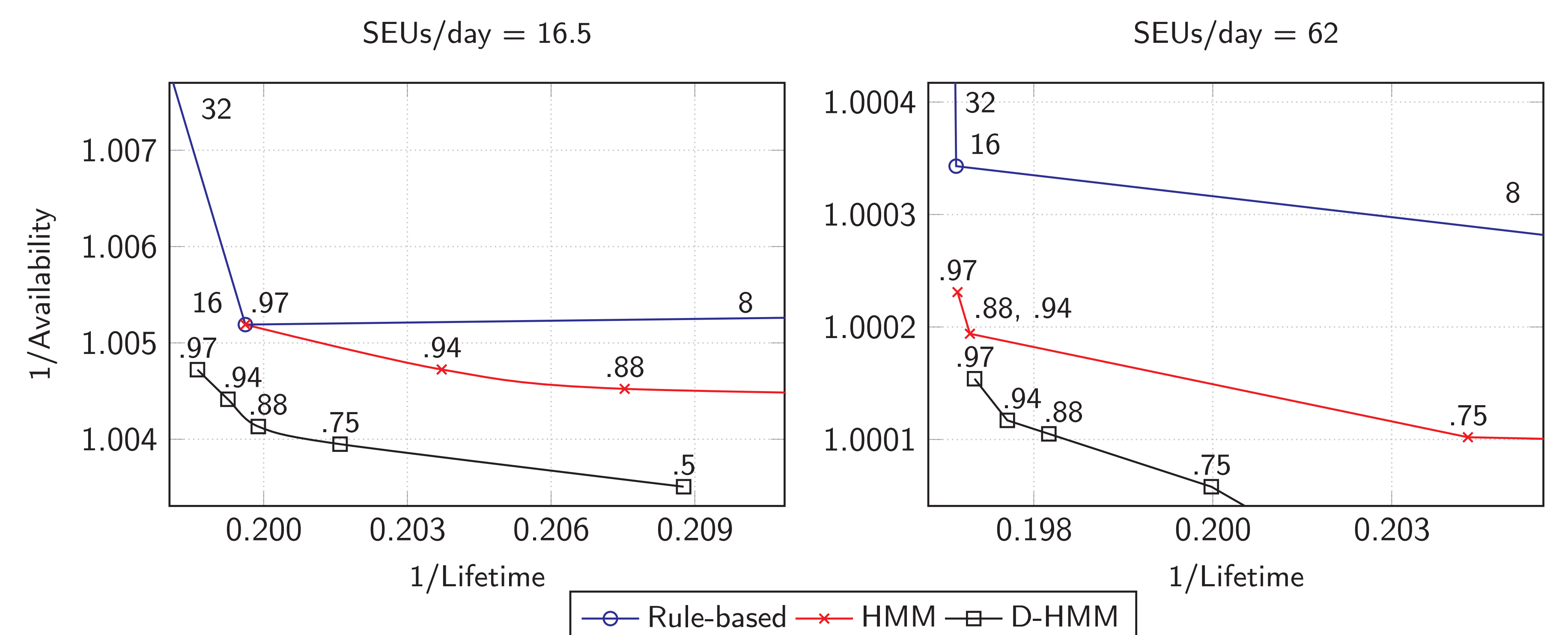
The number of impacts in a scrubbing period of length T is a Poisson rand variable.



6. Simulation Results

Lifetime-availability trade-offs in the form of Pareto Curves.

- Comparison of 3 failure detection models (rule-based, HMM, and D-HMM).
- SEU rates representative of a low-Earth orbit (LEO) and in a highly-elliptical orbit (HEO).
- System redundancy is $N = 10$, and data scrubbing is assumed to take $T = 10$ minutes.



References

- [1] E. Petersen, *Single event effects in aerospace*, 2011.
- [2] S. Gupta, A. Ansari, S. Feng, and S. Mahlke, "Adaptive online testing for efficient hard fault detection," in *2009 IEEE Int. Conf. Comput. Des.* IEEE, Oct. 2009, pp. 343–349.

Acknowledgments

